N186CA Piper PA-32R-301, S/N. 32R-8013046 Concorde Battery STC Installation Instructions for Continued Airworthiness and Maintenance Instructions

...the heart of your aircraft ®

CONCORDE

CONCORDE BATTERY CORPORATION

N186CA

License for use of Supplemental Type Certificate

32R-8013046 PA-32R-301

To Whom It May Concern:

Item 6 11-16526 Squawk Shop order#

Pursuant to Title 49 of United States Code § 44704 (b)(3) (Effective October 19, 1996), CONCORDE BATTERY CORPORATION hereby licenses the holder to use the following listed STC in the noted aircraft

Aircraft Registration No:	N186CA
Registered Owner:	UNIVERSITY OF NEBRASKA
	3835 HOLDREGE ST
	LINCOLN, NEBRASKA 68503-1435
Aircraft Manufacturer and Model No.:	PIPER PA-32R-301
Aircraft Serial No.:	32R-8013046
STC No.:	SA01147WI

- Battery Installed PIN. RG-35AXC
N SIN. 40451540

For

CONCORDE BATTERY CORPORATION

Shelley
Johnson

Digitally signed by Shelley Johnson DN: CN = Shelley Johnson, C = US, O = Concorde Battery Corp., OU = Director of Customer Service Date. 2011.11.03 15:12:34 -07'00'

By: Roxane Montoya

Title: Customer Service

Per: Shelley Johnson

Director of Customer Service

Supplemental Type Certificate

Number SA01147WI

This certificate issued to

CONCORDE BATTERY CORPORATION

1632 South West Street

Suite 4

Wichita, KS 67213

certifies that the change in the type design for the following product with the limitations and conditions therefore as specified hereon meets the aircorthiness requirements of Part 3 of the Civil Air Regulations.

Original Product - Type Certificate Number: * See attached FAA Approved Model List (AML) No. SA01147WI for list of approved models and applicable airworthiness regulations

Make .

Description of Type Design Change:

Installation of Concorde FLA or VRSLA series batteries in the Piper Light Airplanes listed on the attached FAA Approved Model List (AML) No. SA01147WI, dated February 25, 2003, in accordance with Concorde Battery Corporation, Master Drawing List Dwg. No. 5-0157, revision A, dated 12/09/02, or later FAA Approved revision.

Limitations and Conditions:

Compatibility of this design with previously approved modifications must be determined by the installer.

If the holder of this STC agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permit another person to use this certificate to alter the product, the holder shall give the other

This certificate and the supporting data, which is the basis for approval, shall remain in effect until surrendered, suspended, recoked or a termination date is otherwise established by the Administrator of the Federal. Amintion a

Date of application: November 15, 2002

Dalo of issuance. February 25, 2003



Date reissued :

Date amended:

Program Manager

Wichita Aircraft Certification Office

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years,

PAGE 1 of 2 pages This certificate may be transferred in accordance with FAR 21.47

Reference STC # SAO1147WI



CONCORDE BATTERY CORPORATION

March 27, 2006

Mr. Harvey Nero **ACO Manager** FAA (ACO) ACE-115W 1801 Airport Road, Rm 100 Wichita, KS 67209

THE FAA WICHITA AIRCRAFT CERTIFICATION OFFICE, ACE-115W, ACKNOWLEDGES RECEIPT AND ACCEPTS THE APPROVAL CONCURS WITH THE RECOMMENDATION FOR COMMENTS SEE FAALETTER

THIS DATA HAS BEEN ADDED TO OUR FILE

6-9-06 1190

Re. Concorde STC SA01147WI

Dear Mr. Nero:

Concorde would like to make a minor change to the referenced STC. We are adding a new battery type, the RG24-16. This battery is electrically identical to the RG24-15 which has been approved on the STC. The new battery has a slightly different plastic molded outer cover so that it will easily mate with aircraft built to accommodate the CB24-11 battery which has also been approved on the STC.

Concorde has determined that this change is minor within the context of FAR 21.93. We have had this change and Concorde's determination reviewed by Mr. John Newman, DER.

Attached you will find the following documents which need to be updated or added to our STC:

5-0157 Rev B (Master Drawing List, Piper, Light Aircraft, Multiple Models) 5-0156 Rev B (Applicability List, Piper, Light Aircraft, Multiple Models)

RG24-11 Rev B

RG24-15 Rev E

RG24-16 Rev NC

I have also attached copies of the latest revisions to our Instructions for Continued Airworthiness for the two types of batteries that are included in this STC. We would like to also update the STC to include these latest revisions to our recommended maintenance practices.

5-0142 Rev H (ICA Valve Regulated Main Battery) 5-0144 Rev E (ICA Flooded Main Battery)

1632 South West Street, Wichita, KS 67213 • Telephone 316-943-9200 • Fax 316-942-7819

...the heart of your aircraft

CONCORDE BATTERY CORPORATION

Thank you for your assistance.

Sincerely,

John B. Timmons, DN: CN = John

PE

John B. Timmons, PE Vice President Engineering

encl:

5-0157 Rev B (Master Drawing List, Piper, Light Aircraft, Multiple Models) 5-0156 Rev B (Applicability List, Piper, Light Aircraft, Multiple Models) RG24-11 Rev B

RG24-15 Rev E

RG24-16 Rev NC

5-0142 Rev H (ICA Valve Regulated Main Battery) 5-0144 Rev E (ICA Flooded Main Battery)

Reviewed Concorde's determination of minor change per 14CFR21.93.

CONCUR

DO NOT CONCUR

Newman, DER
Newman Consulting LLC
3604 N. Flamingo Ave
Bethany, OK 73008 (405) 789-8148

1632 South West Street, Wichita, KS 67213 ● Telephone 316-943-9200 ● Fax 316-942-7819.

Master Drawing List

			Title
Drawing No.	Rev.	Date	
5-0156 A 3/27/06		3/27/06	APPLICABILITY LIST, PIPER, LIGHT AIRCRAFT, MULTIPLE MODELS
5-0158	NC	11/15/02	INSTALLATION INSTRUCTIONS, PIPER, LIGHT AIRCRAFT, MULTIPLE MODELS
5-0161	NC	11/6/02	INSTALLATION KIT, PIPER AEROSTAR
CB-25	В	1/15/02	BATTERY, LEAD ACID, 12 VOLT
CB-25XC	NC	1/15/02	BATTERY, LEAD ACID, 12 VOLT, PLATINUM SERIES
CB-35A	В	1/15/02	BATTERY, LEAD ACID, 12 Volt
CB-35AXC	NC	1/15/02	BATTERY, LEAD ACID, 12 VOLT, PLATINUM SERIES
CB24-11	В	2/5/02	BATTERY, LEAD ACID, 24 VOLT
CB24-11XC	NC	2/5/02	BATTERY, LEAD ACID, 24 VOLT, PLATINUM SERIES
CB24-11M B		2/5/02	BATTERY, LEAD ACID, 24 VOLT
CB24-11MXC	NC	2/5/02	BATTERY, LEAD ACID, 24 VOLT, PLATINUM SERIES
RG-25	J	11/15/99	BATTERY, VRSLA, 12 VOLT, RG-25 SERIES
RG-35A	В	11/17/99	BATTERY, VRSLA, 12 VOLT, RG-35 SERIES
RG24-11	В	10/4/05	BATTERY, VRSLA, 24 VOLT
RG24-16	NC	1/25/06	BATTERY, VRSLA, 24 VOLT
RG24-11M	А	9/9/98	BATTERY, VRSLA, 24 VOLT
RG24-15	E	10/4/05	BATTERY, VRSLA, 24 VOLT
RG24-15M	С	9/2/99	BATTERY, VRSLA, 24 VOLT
RG24-20	В	1/19/01	BATTERY, VRSLA, 24 VOLT
5-0142	Н	6/24/03	INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, VALVE REGULATED MAIN BATTERY
5-0144	E	10/27/04	INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, FLOODED MAIN BATTERY

evisio	ins		Date	Approved
ev.	Description		12/9/02	JBT
Α	Add drawing 5-0161		3/27/06	JBT
В	Add RG24-16, update drawing rev's RG24-11, RG24-15, ICA's			
ONC	CORDE BATTERY CORPORATION, 2009 San Bernardino Road, West Covins	a, CA 91790		_
UNC	E: Master Drawing List, Piper, Light Aircraft, Multiple Models	DWG NO. 5-0157		REV. B
HL	E: Master Drawing List, 1 1501, 25		LOCUED	CHEET

CHECKED

HRF 11/15/02

DRAWN:

JBT 11/5/02

CAGE CODE:

\$3017

APPROVED

JBT 11/15/02

SHEET:

1 OF 1

ISSUED

NS 3/27/06

		Applic	ability List	Type	Certification Basis	Model No.	Serial No.
Туре	Certification	Model No.	Serial No.			PA-22S-160	22-2378, 22-2425 and up
Certificate	Basis		LO 4 and UD			PA-22-108	22-8000 and up
TC A-780	CAR 3	PA-12	12-1 and up	C 1A10	CAR 3I	PA-23 APACHE	23-1 through 23-182
		PA-12S	12-1 and up			PA-23-160	23-1177, 23-1183 through 23-2046
TC 1A2	CAR 3	PA 18 PA-19 (Army L-18C)	18-1 through 18-7632 19-1 through 19-3			PA-23-250 (Navy UO-1)	27-1 through 27-4916, 27-7304917 through 27-7554168
		PA-18 "125" (Army L-21B)				PA-23-235	27-205 through 27-622
		PA-18 "135" (Army L-21B)	reacted Secret			PA-E23-250 AZTEC	27-2505 through 27-4916,
		PA-18A	Tolerand Greek Tolerand			1 A-220-200 7/21/20	27-7304917 through 27-7554168
		PA-18A "135"	Annual Control of the			PA-23-250	27-7654001 through 27-8154030
		PA-18 "105" (Special)		C 1A15	CAR 3	PA-24 COMANCHE	24-1 through 24-3687
		PA-18 PA-18S				PA-24-250	24-1, 24-103 through 24-3687
		PA-19S				PA-400	26-2 through 26-148
		PA-18S "125"				PA-24-260	24-3642, 24-4000 through 24-5034
				TC 2A8	CAR 3	PA-25 PAWNEE	25-1 through 25-731
		PA-18S "135" PA-18AS "125"		WHEN THE		PA-25-235	25-02, 25-2000 through 25-5521 and 25-730552 through 25-8156024
		PA-18AS "135"PA-18S "105"	40.0774.40.0704.14			PA-25-260	25-4415 through 35-5521 and 25-730522 through 25-8156024
		PA-18 "150"	18-3771, 18-3781,through 18-9015	50,0040	CAR 8.10(b)	PA-25 PAWNEE	25-1 through 25-731
		PA-18A "150"	18-7309016 through 18-8309025	FC 2A10	CAR 6. 10(b)	PA-25-235	25-02, 35-2000 through 25-5521 and
TO AD 7	CAD 0.40(h)	PA-18S "150"	1809001 through 1809113			17/120/200	25-7305522 through 25-8156024
TC AR-7	CAR 8.10(b)	PA-18A	18-677, 18-770, and 18-262 through 18-3779	MINER TO		PA-25-260	25-4415 through 25-5521 and 25-7305522 through 25-8156024
		PA-18A "135"	18-3786 through 18-9105 and 18-7309016 through	TC 2A13	CAR 3	PA-28-160	28-03, 28-1 through 28-4377 and 28-1760A
		PA-18A "150"	18-8309025			PA-28-150	28-03, 28-1 through 28-4377 and
TC 1A4	CAR 3	PA-20	20-1 and up			FA-20-100	28-1760A
		PA-20 "115"	20-1 and up			PA-28-180	28-03, 28-671 through 28-5859 and 28-7105001 through 28-7205318
		PA-20S				71 000 100	28-03, 28-1 through 28-1760 and
		PA-20S "115"				PA-28S-160	28-1760A
		PA-20S "135"				PA-28-180	28-671 through 28-5859 and
		PA-20 "135"	20-877 and up				28-7105001 through 28-7105234
TC 1A6	CAR 3	PA-22	22-1 and up			PA-28-235	28-10001 through 28-11378 and 28-7110001 through 28-7210020
		PA-22-135	22-534 and up			PA-28-140	28-20001 through 28-26946 and
		PA-22S-135	22-534 and up				28-7125001 through 28-7725290
		PA-22-150	22-2378, 22-2425 and up			PA-28R-180	28R-30002 through 28R-31270 and 28R-7130001 through 28R-7130013
		PA-22S-150	22-2378, 22-2425 and up				20K-/ 130001 tillough 20K-/ 100010
		PA-22-160	22-2378, 22-2425 and up				

TITLE A PLANTAGE AND	ITLE: Applicability List, Piper, Light Aircraft, Multiple Models	DWG. NO. 5-0156	REV. A
TITLE: Applicability List, Piper, Light Aircraft, Multiple Models CAGE CODE: DWG. NO. 5-0	66 REV. AGE CODE:		SHEET: 2 OF 6
63017	1 OF 6 3017		

Type Certificate	Certification Basis	n Model No.	Serial No.	Type ertificate	Certification Basis	Model No.	Serial No.	
		PA -28R-200	28R-35001 through 28R-35820 and 28R-7135001 through 28R-7135229	Fruncate	Dasis	PA-32S-300	32S-15, 32S-40000 through 32S-40974 and	
		PA-28R-200	28R-7235001 through 28R -7635545				32S-7140001 through 32S-7240137	
		PA-28-180	28E-13,			PA-32R-300	32R-7680001 through 7880068	
			28-7305001 through 28-7505260			PA-32RT-300	32R-7885002 through 32R-7985106	
		PA-28-235	28E11, 28-7310001 through 28-7710089			PA-32RT-300T	32R-7787001, 32R-7887002 through 32R-7987126	
		PA-28-151	28-7415001 through 28-7715314			PA32R-301	32R-8013001 through 32R—8613006, 3213001 through 3213028 and 3213030 through 3213041	
		PA-28-181	28-7690001 through 28-8690056, 28-8690061, 28-8690062, and 28-90001 through 2890205			PA-32R-301	3213029, 3213042 through 3213103 (14 volt) 3246001 through 3246017 (14 volt) and 3246018 and up (28 volt)	
		PA-28-181	28-90206 through 28-90231 and 2843001 and up			PA -32R-301T	32R-8029001 through 32R-86-32009 and 3229001 through 3229003	
		PA-28-161	28-7716001 through 28-8216300 and 2841001 through 2841365			PA-32-301	32-8006002 through 32-8606023 and 3206001 through 3206019	
		PA-28-161	28-8116001 through 28-8616057 and 2816001 through 2816109			PA-32-301T	32-8024001 through 32-8424002	
		PA-28-161	2816110 through 2816119 and			PA-32R-301T	3257001 and up	
		17/20 101	2842001 and up	C A9SO	FAR 23	PA-36-285	36-7360001 through 36-7660135	
		PA-28R-201	28R-7737002 through 28R-7837317,	A SECTION AND A SECTION AND ASSESSMENT OF THE PARTY OF TH		PA-36-300	36-7560001 through 36-8160023	
			2837001 through 2837061, and 2844001 and up 28R-7703001 through 28R-7803374 and			PA-36-375	36-7802001 through 36-8302025	
		PA-28R-201T		C A10SO	FAR 23	PA-36-285	36-7360001 through 36-7660135	
			2803001 through 2803012	l and		PA-36-300	36-7560001 through 36-8160023	
		PA-28-236	28-7911001 through 28-8611008,			PA-36-375	36-7802001 through 36-8302025	
			2811001 through 2811050, and 2845001 and up	C A18SO	FAR 23	PA-38-112 TOMAHAWK	38-78A0001 through 32-82A0122	
		PA-28RT-201	28R-7918001 through 28R-7918267	C A19SO	FAR 23	PA-44-180 SEMINOLE	44-7995001 through 44-8195026	
		PA-28RT-201	28R-8018001 through 28R-8218026	10000		PA-44-180 SEMINOLE	44-95001 through 44-95013,	
		PA-28RT-201T	28R-7931001 through 28R-8631005 and 2831001 through 2831038			PA-44-180T TURBO	44-96001 and up 44-8107001 through 44-8207020	
		PA-28-201T	28-7921001 through 28-7921095	C A25SO	EAD 00	SEMINOLE PA-46-310P MALIBU	40.0400004.11	
C A20SO	CAR 3 / FAR 23	PA-31 NAVAJO	31-2 through 31-900, 31-7300901 through 31-8312019	C A2550	FAR 23	PA-40-310F WALIBU	46-8408001 through 46-8408087 46-8508001 through 46-8508109 46-8608001 through 46-8608067	
		PA-31-300 NAVAJO	31-2 through 31-900, 31-7300901 through 31-8312019			PA-46-350P MALIBU MIRAGE	4608001 through 4608140 4622001 through 4622200	
		PA-31-350 CHIEFTAIN & T-1020	31-5001 through 31 5004	FC 4454	045.0	DA 20 TIMINI COMANICUE	4636001 and up	
		PA-31-325 NAVAJO C/R	31-7305005 through 31-8553002	TC A1EA	CAR 3 also	PA-30 TWIN COMANCHE	30-1 through 30-2000	
C A3SO	CAR 3	PA-32-260	31-7400990, 31-7512001 through 8312019		FAR 23	PA-39 TWIN COMANCHE	39-1 through 39-162	
			32-03, 32-04, 32-1 through 32-1297 and 32-7100001 through 32-7800008	C A7SO	FAR 23	PA-40 TWIN COMANCHE PA-34-200 SENECA	40-7400002 34-ER, 34-7250001 through 34-7450220	
		PA-32-300	32-15, 32-21, 4000 through 32-40974 and 32-7140001 through 32-7940290			PA-34-200T SENECA II	34-750001 through 34-8170092	

TITLE: Applicability List, Piper, Light Aircraft, Multiple Models CAGE CODE:	DWG. NO. 5-0156	REV. A	TLE: Applicability	List, Piper, Light Aircr	aft, Multiple Models	DWG. NO. 5-015	66 REV. A
63017	5775. NO. 3-0136	CHFET:	AGE CODE: 3017				SHEET: 4 OF 6

Type Certificate	Certification Basis	Model No.	Serial No.
		PA-34-220T SENECA III	34-8133001 through 34-8633031 (14V) 3433001 through 3433172 (14V) 3448001 through 3448037 (28V)
		PA-34-220T SENECA IV	3448038 through 3448079 3447001 through 3447029
		PA-34-220T SENECA V	3449001 and up.
A17WE	FAR 23	PA-60-600	60-0001 thru 60-0014, 60-0015-26 thru 60-0056-126, 60-013 057 thru 60-0933-8161262, 60-8261001 and subsequent.
		PA-60-601	SN 61-0001 thru 61-0002-19 thru 61-0070-129, 61-0132-07- thru 61-0880-8162157, 60-82620001 and subseqent and 62- 0001-013.
		PA-60-601P	61P-0157-001 thru 61P-0860-8163456, 60-8263001 and subsequent.
		PA-60-602P	62P-0750-8165001, 62(-0861-8165002 thru 62P-0932- 8165056, 60-8265001 and subsequent.
		PA-60-700P	60-8223001, 60-8423001 and up.

-	Piper P/N	Concorde P/N	Ampere-Hour (AH) Capacity at the 1 hour rate (C1)	Weight (lbs.)
	To replace the Piper 12 volt battery P/N 450-030	Concorde CB-25 Concorde CB-25XC Concorde RG-25 Concorde RG-25XC	20 AH 25 AH 22 AH 24 AH	22 lbs. 24.5 lbs. 22.75 lbs. 23.5 lbs.
3(To replace the Piper 12 volt battery P/N 450-022	Concorde RG-25	22 AH	22.75 lbs.
1	To replace the Piper 12 volt battery P/N 450-035	Concorde CB-35A Concorde CB-35AXC Concorde RG-35A Concorde RG-35AXC	29 AH 34 AH 29 AH 33 AH	28.5 lbs. 32.5 lbs. 29.5 lbs. 32 lbs.
	To replace the Piper 24 volt manifold battery P/N 450-101	Concorde CB24-11M Concorde CB24-11MXC Concorde RG24-11M Concorde RG24-15M	10 AH 15 AH 11 AH 13.6 AH	28 lbs. 31 lbs. 26.5 lbs. 29.5 lbs.
		The Concorde RG-24-110 required. Cap and stow or 43HT4-48 spacer or equivocompensate for height difference.	remove the existing mani valent ¼ inch spacer on ho	fold vent line. Install NAS
	To replace the Piper 24 volt battery P/N 450-027 or 450-032	Concorde CB24-11 Concorde CB24-11XC Concorde RG24-11 Concorde RG24-15 Concorde RG24-16	10 AH 15 AH 11AH 13.6 AH 13.6 AH	27 lbs. 31 lbs. 26.5 lbs. 29.5 lbs. 29.5 lbs.
	To replace the Piper 24 volt battery P/N 450-028, or 450-031	Concorde RG24-20	19 AH	42 lbs.
	To replace the Piper 24 volt battery P/N 450-029	Concorde RG24-20	19 AH	42 lbs.

The purpose of this report is to provide instructions for the replacement of the original equipment flooded lead acid battery with a Concorde Battery Corporation flooded lead acid or Value Battery Corporation floo with a Concorde Battery Corporation flooded lead acid or Valve Regulated Sealed Lead Acid (VRSLA) Battery in Piper

DESCRIPTION

The Concorde flooded lead acid or VRSLA RG series batteries referenced here are direct replacements for the flooded lead acid batteries originally installed in those girareft. No market is referenced here are direct replacements for the flooded lead acid or VRSLA RG series batteries referenced here are direct replacements for the flooded lead. acid batteries originally installed in these aircraft. No mechanical or electrical modifications of the aircraft are required to install the Concorde battery. The Concorde VRSLA battery requires no venting.

A supplemental kit is required for the installation of the RG24-20 battery in place of Piper 450-029.

The Concorde batteries are compatible to all electrical specifications and operating procedures as specified in the Pilots Operating Handbooks and should be operated in exactly the same manner as the original battery. Any references to adding water can be ignored if the VRSI A battery is selected. Inspection and the original battery. water can be ignored if the VRSLA battery is selected. Inspection and maintenance can be performed in any shop by reference to the latest revision of Concorde Battery Corp., Instructions for Continued Airworthiness (ICA).

The Concorde VRSLA Battery can be safely recharged in the aircraft by following the Concorde Instructions for Continued Airworthiness (ICA) that are part of this document

The Concorde batteries offer the operator a choice of capacities to suit the particular mission.

TITLE: Applicability List, Pipe	, Light Aircraft, Multiple Models	particular mission.	
CAGE CODE: 63017	worders woders	DWG. NO. 5-0156	REV. A
			SHEET: 5 OF 6

ecked by John Newman, DER Electrical, Newman Consulting LLC, Bethany, OK.. 73008 (405) 789-8148

vision	IS							
lev.	Description						Date	Approved
Α	Add RG24-16 battery							JBT
ITLE	TLE: Applicability List, Piper, Light Aircraft, Multiple Models DWG. NO. 5-01						156	REV. A
AGE 3017	CODE:	DRAWN: DGG 11/5/02	CHECKED HRF 11/15/02	APPROVED JBT 11/15/02		ISSUED NS 3/27/	06	SHEET: 6 OF 6

PIPER INSTALLATION INSTRUCTIONS

1.0 REMOVAL

- 1.1 Assure that the "Battery Master Switch" is "OFF".
- 1.2 Gain access to the battery compartment and loosen the battery hold downs, if not secured by the battery box lid.
- 1.3 Disconnect the vent tubes if installed on the battery. Cap and stow if replacement battery is not manifold type.
- 1.4 Disconnect the battery cables. Always disconnect the Negative cable first and reconnect it last for safety reasons.
- 1.5 Remove the battery and PROPERLY DISPOSE OF IT.
- 1.6 Clean the battery compartment and inspect the cables and connector for airworthiness. Refer to applicable service manuals instructions for cleaning, repairs and/or preventive maintenance operations on battery compartment.

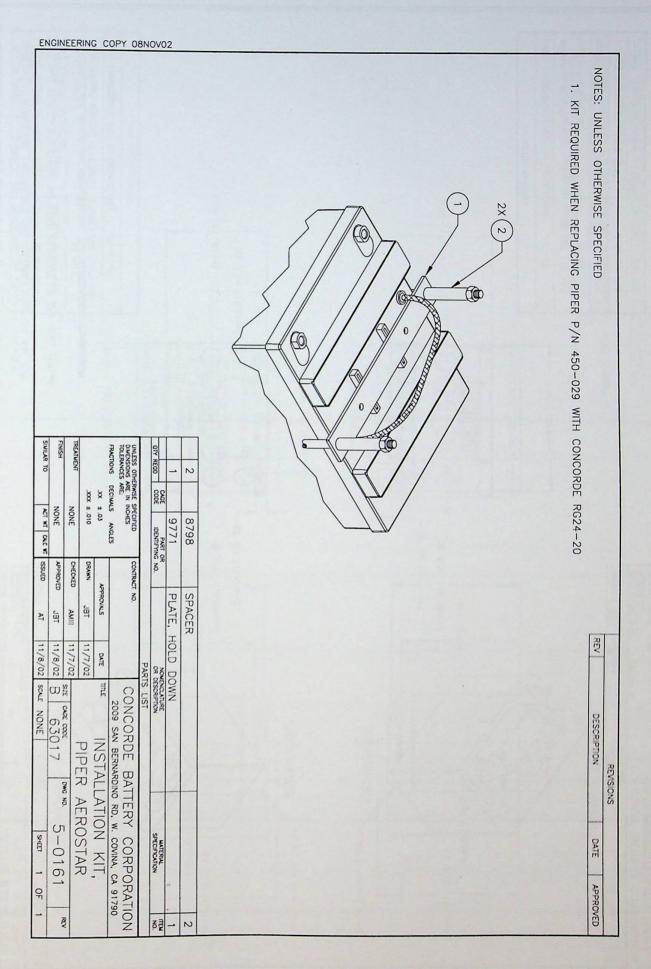
2.0 INSTALLATION

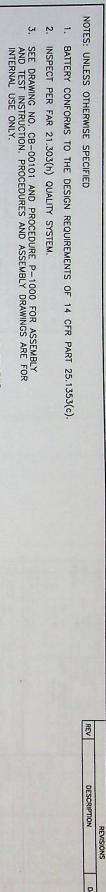
- 2.1 Install the fully charged Concorde Battery in the battery compartment. Engage the battery hold down if not secured by the battery box lid, tighten and safety wire. Reconnect battery cables, Negative cable last.
- 2.2 Installation of the Concorde RG-25 to replace the Piper 450-022 requires removing and stowing the drain tubing. Drai tubing is not required for a VRSLA battery.
- 2.3 Installation of the Concorde RG24-20 to replace the Piper 450-029 requires the supplemental installation kit shown in Concorde Drawing 5-0161.
- 2.4 Make entry into the aircraft records to show the accomplishment of this battery installation, including a change of the aircraft weight and balance if the replacement battery has a weight different from the battery being replaced. Refer to the flight manual or appropriate document to determine station number of battery location.

3.0 MAINTENANCE

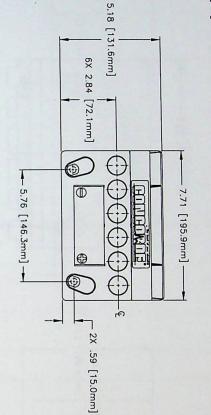
- 3.1 The RG series battery requires no maintenance other than inspection of connections for security. This battery require no added water.
- 3.2 Refer to the latest revision of Concorde Battery Corporation Instructions for Continued Airworthiness (ICA) for periodic inspection requirements and test procedures for the battery selected. The ICA is available on the Concorde battery web si at www.concordebattery.com

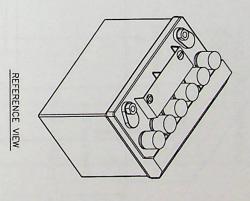
CONCORDE BAT	TERY CORPORA	TION, 2009 San D.	dino Road, West Covina, CA		
TITLE: Installation Ins Models	tructions, Piper, Light	DWG NO. 5-0158	REV. No		
CAGE CODE: 63017 DRAWN: JBT 11/5/02		CHECKED HRF 11/15/02	APPROVED JBT 11/15/02	ISSUED AT 11/15/02	SHEET: 1 OF 1

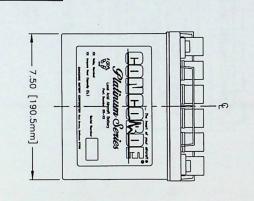


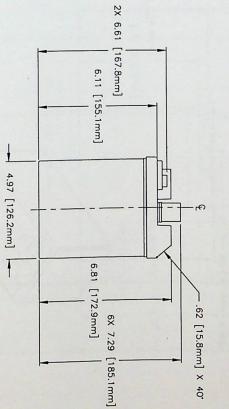


- TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS M8 HEX HEAD SCREWS AND CONICAL WASHERS.
- BATTERIES ARE RATED IAW INTERNATIONAL ELECTROTECHNICAL COMMISSION STD. 952-1 UNLESS OTHERWISE SPECIFIED.
- ALL DIMENSIONS ARE IN INCHES[MM].









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るはくとととうのつい	BATTERY, LEAD ACID,	ONCORDE BATTERY CORPORATION SAN BERNARDINO RD, W. COVINA, CA 91790		SPECIFICATION	
20		NO		50	

POWER RATING AT -30°C (-22°F) POWER RATING AT -18°C (0°F) POWER RATING AT 23°C (74°F)

lpr 371 A, lpp 447 A

lpr 738 A, lpp 895 A lpr 513 A, lpp 625 A

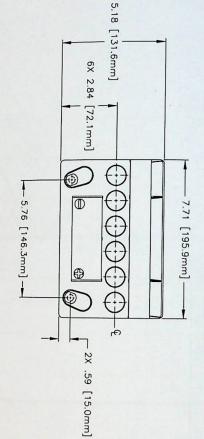
24.5 LBS [11.2 KG] 25 AH

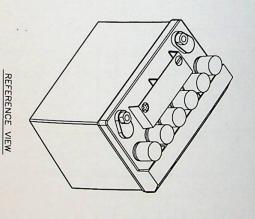
MAXIMUM WEIGHT
RATED CAPACITY, C1

PART NUMBER

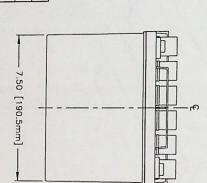
CAPACITY RATINGS

- SEE DRAWING NO. CB-00101 AND PROCEDURE P-1000 FOR ASSEMBLY AND TEST INSTRUCTION. PROCEDURES AND ASSEMBLY DRAWINGS ARE FOR INTERNAL USE ONLY.
- TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS M8 HEX HEAD SCREWS AND CONICAL WASHERS.
- ALL DIMENSIONS ARE IN INCHES[MM]. BATTERIES ARE RATED IAW INTERNATIONAL ELECTROTECHNICAL COMMISSION STD. 952-1 UNLESS OTHERWISE SPECIFIED.

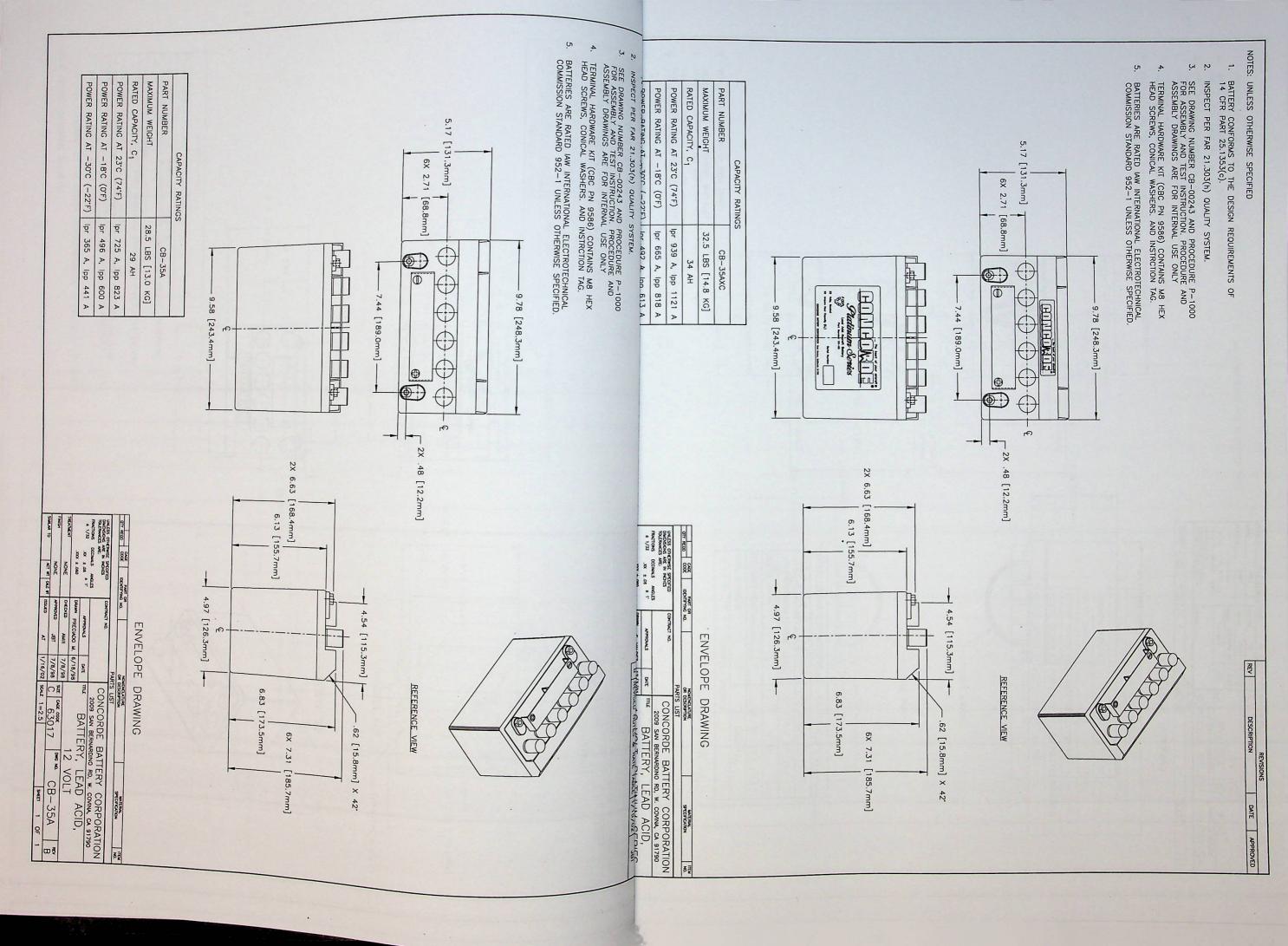


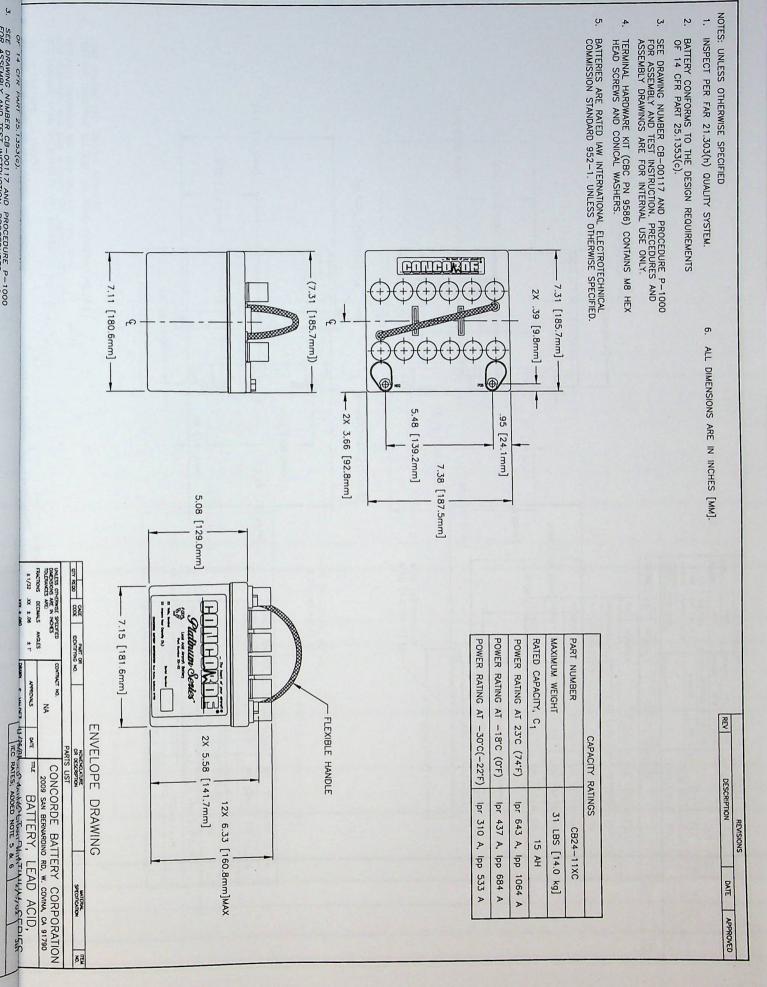


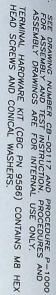
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(3.č						CAPACITY RATINGS	
lpr	lpr	lpr				S	
230	340	525		22 L			
.>	A	,	20 AH	BS	CB-25		
Ipp	ipp	lpp	AH	[10	25		
270	lpr 340 A, lpp 395 A	lpr 525 A, lpp 635 A		22 LBS [10 KG]			
D	A	A					



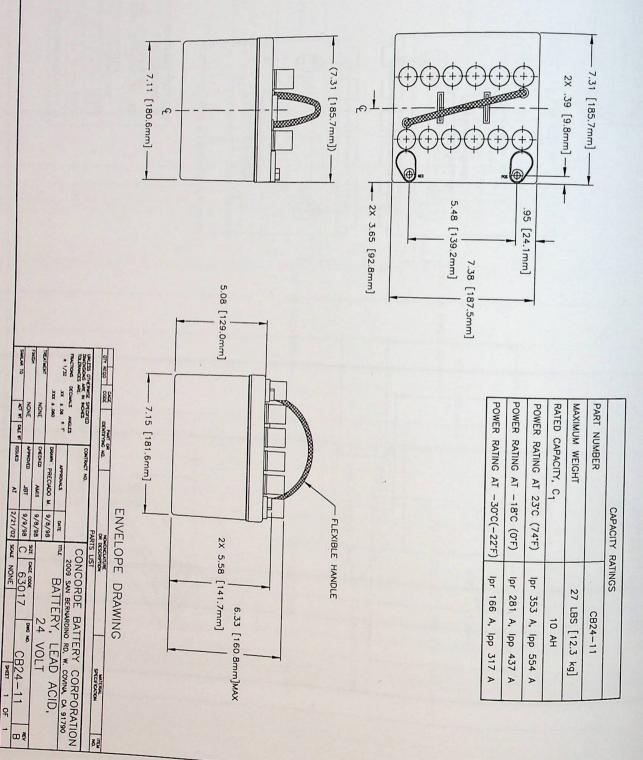
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1/16/02	4/30/99	4/30/99	A 2/16/99			PAR	S NO	4.97 [126.2mm] ENVELOPE	
	C 63017 CB-25 BY	IZ VULI	יייינייי, רנויס לכוס,	BATTERY FAD ACID	CONCORDE BATTERY CORPORATION 2009 SAN BERNARDINO RD, W. COVINA, CA 91790		OR DESCRIPTION SPECIFICATION NO.	4.97 [126.2mm] -	.62 [15.8mm] × 40. 6x 7.29 [185.1mm]







ALL DIMENSIONS ARE IN INCH [MM].

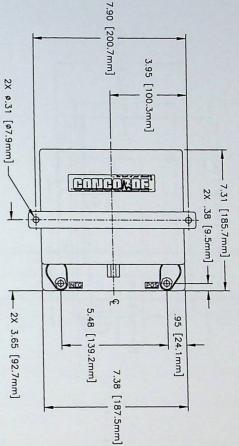


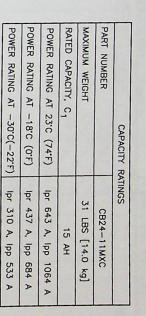
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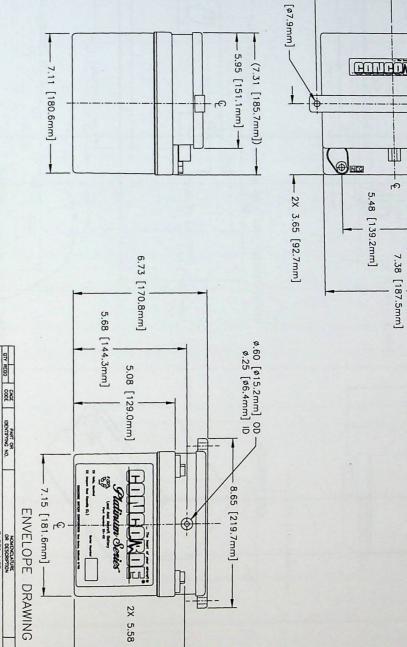
1. INSPECT PER FAR 21.303(h) QUALITY SYSTEM. 6. ALL DIMENSIONS ARE IN INCHES [MM].

- BATTERY CONFORMS TO THE DESIGN REQUIREMENTS OF 14 CFR PART 25.1353(c).
- SEE DRAWING NUMBER CB-00118 AND PROCEDURE P-1000 FOR ASSEMBLY AND TEST INSTRUCTION. PROCEDURE AND ASSEMBLY DRAWINGS ARE FOR INTERNAL USE ONLY.

 TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS MB HEX HEAD SCREWS, CONICAL WASHERS AND INSTRUCTION TAG.
- BATTERIES ARE RATED IAW INTERNATIONAL ELECTROTECHNICAL COMISSION STANDARD 952-1 UNLESS OTHERWISE SPECIFIED.







[141.7mm]

SEE DRAWING NUMBER CB-00118 AND PROCEDURE P-1000 FOR ASSEMBLY AND TEST INSTRUCTION. PROCEDURES AND ASSEMBLY DRAWINGS ARE FOR INTERNAL USE ONLY.

TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS MB HEX HEAD SCREWS AND CONICAL WASHERS.

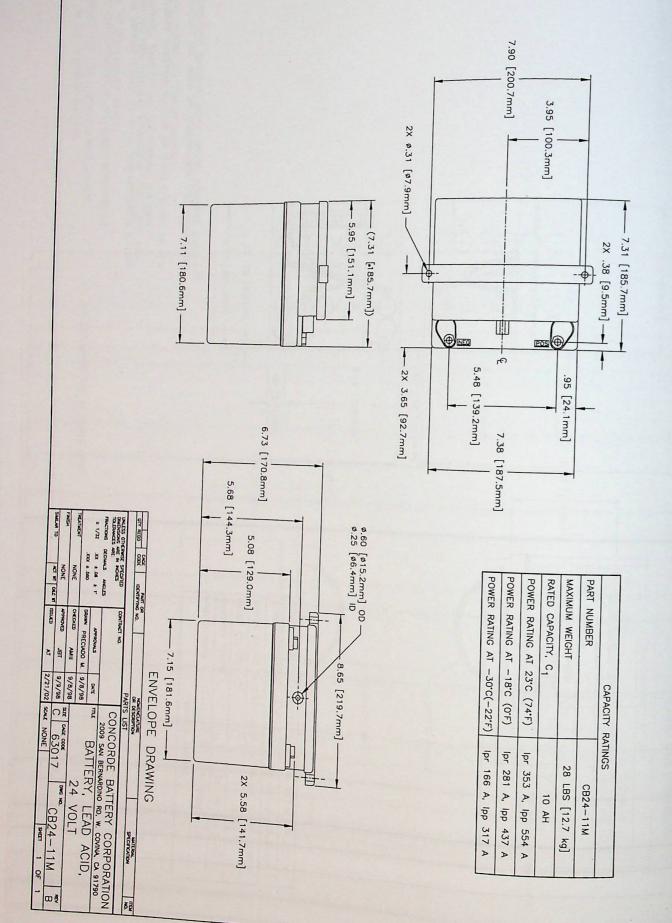
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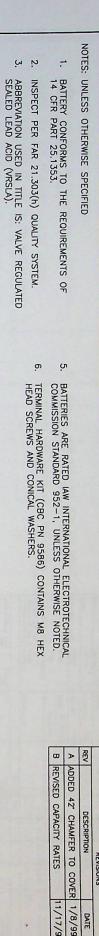
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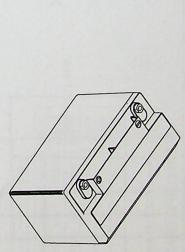
CONCORDE BATTERY CORPORATION
2009 SAN BERNARDINO RD, W. COVINA, CA 91790
BATTERY, LEAD ACID,
S WLDEZ JIJ (2012 - 24 MINES J. T. BURT (NJ. M.). SEPJES

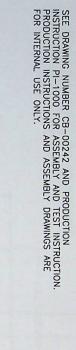
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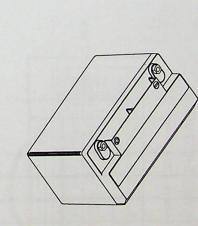
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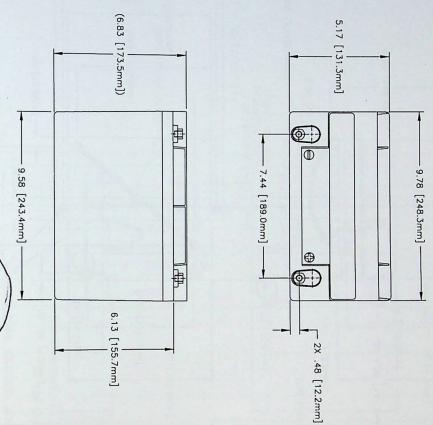


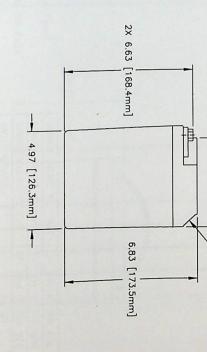


REFERENCE VIEW

4.54 [115.3mm]

- .62 [15.8mm] X 42"





LVE	S ALL		8°C		C						
ALVE REGULATED SEALED	ITY SYSTEM.	the part of the state of the st	8.C Ipr 381 A, Ipp 124/ A Ipr /18 A, Ipp 129/ A	100	lpr 898 A, lpp 1323 A/ lpr 1016 A, lpp 1343 A	29 AH /	1	בטים במם [ים: יים]	29 5 IBS [13 4 KG]	RG-35A	
7. BATTERIES ARE RATED IAW INTERNATIONAL ELECTROTECHNICAL COMMISSION STANDARD 959-11 LINESCO CTATERING PROPRIESTO PROPRIE	ARE FOR INTERNAL USE ONLY.	DDD COLOR	Ipr / 18 A, Ipp 129/ A	120	lpr 1016 A, lpp 1343 A	33 AH		יייי וייי	/ 32 IBS [14.5 KG]	/ RG-35AXC	
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NICAL		DELLEG		200	CONTRACT NO.		IDENTIFYING NO.	PART OR			
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FROM 23.0 TO 23.5	FROM 26 AH TO 24 AH	2011	BATTERY, VRSLA, 12 VOLT,	2009 SAN BERNARDINO RD, W. COVINA, CA 91790	CONCORDE BATTERY CORPORATION	PARTS LIST	OR DESCRIPTION	HOMENCIATURE		ENVELOPE DRAWING	
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JBT	181		17.	1790	ATION		30	2			

RECOMMENDED SUBSTITUTE FOR OBSOLETE PN RG 25M (AC 54).

MAXIMUM WEIGHT

RATED CAPACITY, C1

POWER RATING AT 23°C

POWER RATING AT -18°C

POWER RATING AT -18°C

POWER RATING AT -18°C

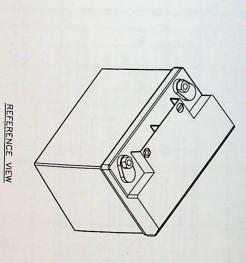
INSPECT PER FAR 21.303(h) QUALITY SYSTEM.

3. ABBREVIATION USED IN TITLE IS: VALVE REGULATED SEALED

BATTERIES ARE RATED IAW INT COMMISSION STANDARD 952-1

PART NUMBER

TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS MB HEX HEAD SCREWS AND CONICAL WASHERS.



6.81 [172.9mm] 6.11 [155.1mm]	A MO A 1.25 INJERNAL THREAD	5.76 [146.3mm]	7.71 [195.9mm] ———————————————————————————————————
	HERNAL THREAD	nm]	

	2X 6.61
	2X 6.61 [167.8mm]
4.97 [126.2mm]	
1	

RG-25 22.75 LBS [10.3 KG] 22 AH Ipr 508 A, Ipp 1278 A Ipr 250 A, Ipp 1254 A Ipr 145 A, Ipp 854 A Ipr 257 A, Ipp 846 A Ipr 257 A, Ipp 854 A

7.50 [190.5mm] —

	lpr	lpr	ipr		23	
	lpr 257 A, lpp 494 A	lpr 375 A, lpp 536 A	lpr 597 A, lpp 846 A	24 AH	23.5 LBS [10.7 KG]	NO-23XC
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NOTES: UNLESS OTHERWISE SPECIFIED SEE DRAWING NUMBER CB-00350 AND PROCEDURE P-1000 FOR ASSEMBLY AND TEST INSTRUCTION. PROCEDURES AND ASSEMBLY DRAWINGS ARE FOR INTERNAL USE ONLY. INSPECT PER FAR 21.303(h) QUALITY SYSTEM. ABBREVATION USED IN TITLE IS: VALVE REGULATED SEALED LEAD ACID (VRSLA). BATTERY CONFORMS TO THE DESIGN REQUIREMENTS OF 14 CFR PARTS 23.1353, 25.1353(c), 27.1353, 29.1353(c). ROPE LIFTING HANDLE 6. 7.31 [185.7mm] BATTERIES ARE RATED IAW INTERNATIONAL ELECTROTECHNICAL COMMISSION STANDARD 60952-1, UNLESS OTHERWISE NOTED. TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS M8 HEX HEAD SCREWS AND CONICAL WASHERS. 5.48 5.08 [129.0mm] .95 [24.1mm] 2X .38 [9.5mm] [139.2mm] 2X 7.38 [187.5mm] 5.58 [141.7mm] 5.68 [144.2mm] PART NUMBER

MAXIMUM WEIGHT

RATED CAPACITY, C1

POWER RATING AT 23°C

POWER RATING AT -18°C

POWER RATING AT -30°C RG24-16
29.5 LBS [13.4]
13.6 AH
1pr 620 A, 1pp 930 A
1pr 320 A, 1pp 615 A
1pr 180 A, 1pp 390 A

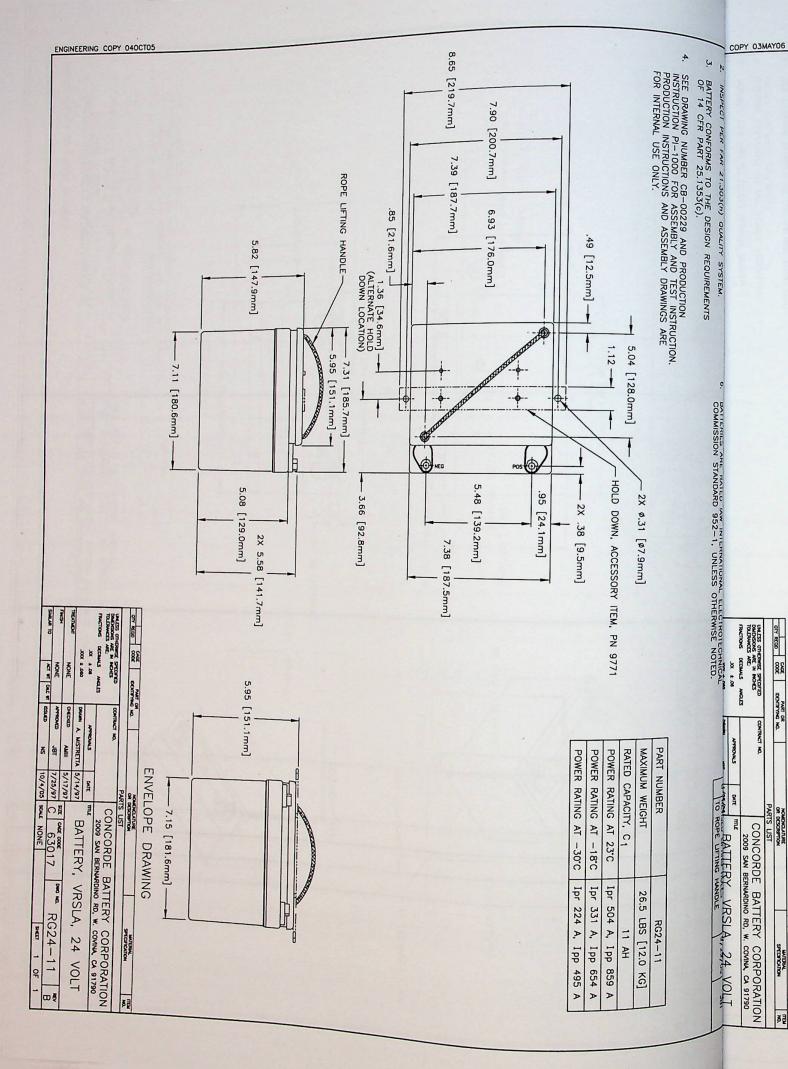
7.11 [180.6mm]

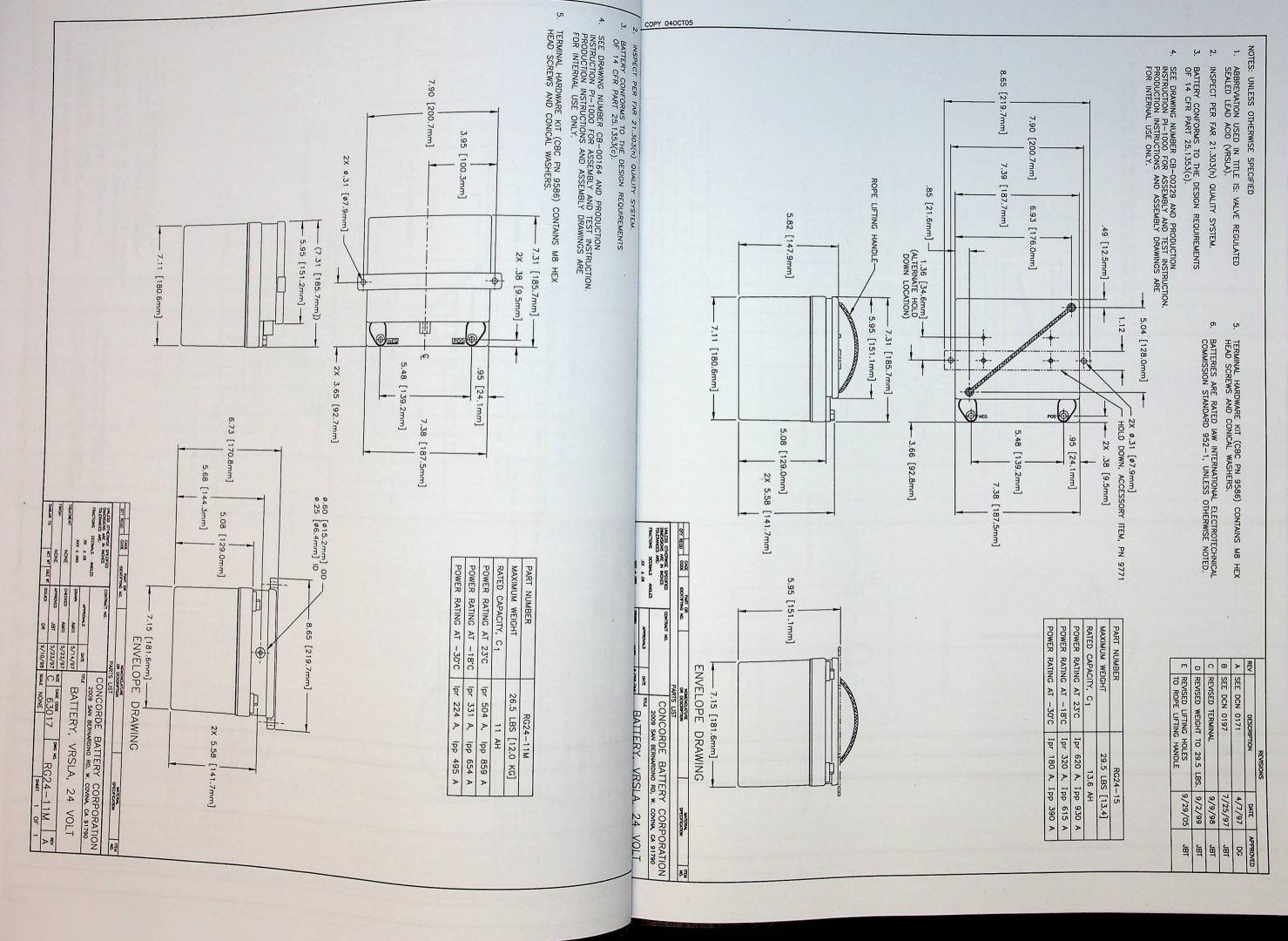
OTY RECOO COOK

ENVELOPE DRAWING

CONCORDE BATTERY CORPORATION
2009 SAN BERNARDINO RD, W. COVINA, CA 91790

SAECHUCYLON TWILDRAY





5/13/97 BATTERY, VRSLA, 24 VOLT
5/23/97 C 630017 C 63017 RG24-11M 9/10/98 sour NONE seer 1 0F 1

BATTERY, VRSLA, 24 VOLT

CONCORDE BATTERY CORPORATION 2009 SAN BERNARDING RD. W. COVINA, CA 91790

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CONCORDE BATTERY CORPORATION
2009 SAN BERNARDINO RD, W. COVINA, CA 91790

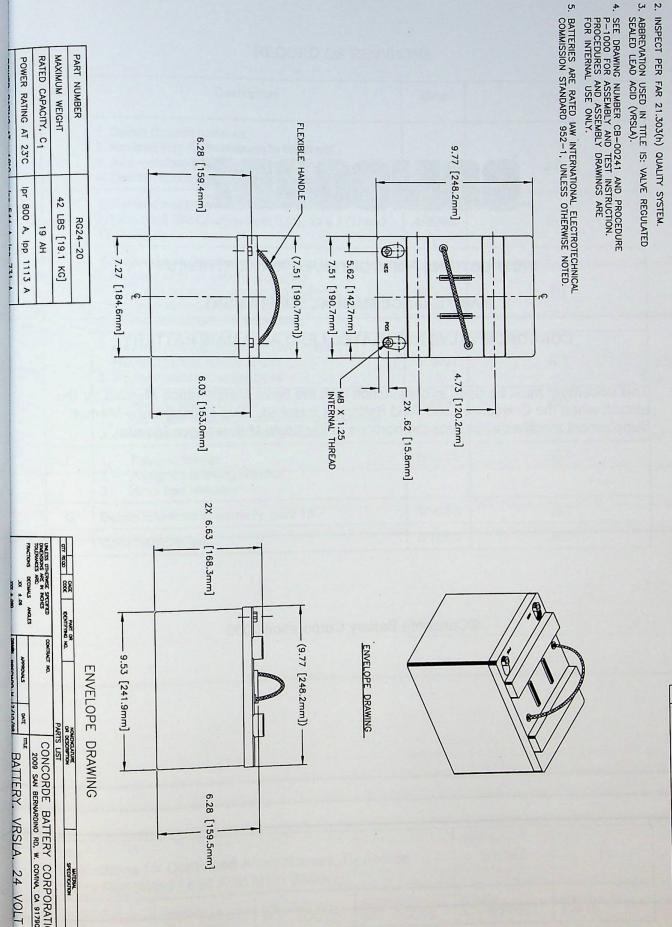
6.28 [159.5mm]

TERMINAL HARDWARE KIT (CBC PN 9586) CONTAINS M8 HEX HEAD SCREWS AND CONICAL WASHERS.

NOTES: UNLESS OTHERWISE SPECIFIED

1. BATTERY CONFORMS TO THE DESIGN REQUIREMENTS OF 14 CFR PART 25.1353(c).

A INC 8242 REV E CHANGES
B REMOVED SLEEVE, REV. HT.
OF CONTAINER (WAS) 6.250 6/7/99 1/19/01 APPROVED JBT JBT





INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

MAINTENANCE MANUAL SUPPLEMENT

CONCORDE VALVE REGULATED LEAD ACID MAIN BATTERY

This document must be used in conjunction with the basic Maintenance Manual for the aircraft when the Concorde Lead Acid Battery is installed. This Maintenance Manual Supplement modifies/augments the portions of the basic Maintenance Manual.

©Concorde Battery Corporation 2005

RECORD OF REVISIONS

Rev. No.	Description	Date	Appd
Α	Delete flooded batteries Revised inspection interval Add mandatory removal interval Deleted flooded battery facility requirement	2/16/99	JBT
В	1. Revised inspection interval 8 2. Removed 282 Charger from 8.(b). 10.b ii (1) and 13.C i (a)	4/20/99	JBT
С	Revised installation instructions Revised note on experience rating Add note on deeply discharged batteries Revised servicing instructions Added non scheduled inspections	9/5/00	JBT
D	Revised installation and removal notes Revised torque specifications Revised battery charger types	6/18/01	JBT
E	Rev cap test to C1 rate Rev cutoff voltage for conditioning procedure Add warning re conditioning charge	1/30/02	JBT
F	Revised format Assigned Drawing Number Minor text revisions	6/20/02	JBT
G	Delete reference to warranty, para 13	1/16/03	JBT
Н	Completely revised	6/12/03	JBT

Instructions for Valve Regulate	r Continued Ai	rworthiness, C	concorde	5-0142	Н
CAGE CODE		CHECKED EFK 6/20/02	APPROVED	ISSUED AT 6/24/03	SHEET 2 of 10

CONTENTS

1.Scope	4
2.Purpose	4
3.Application	
4.Definitions	
5.Precautions:	4
6.Airworthiness Limitations	5
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9.Troubleshooting:	8
10.Servicing Discharged Batteries:	8
11.Repair / Replacement:	9
12.Facilities:	9
13.Storage Limitations:	9
14.Disposal:	10

Concorde Valve Regulated Lead Asides		
Concorde Valve Regulated Lead Acid Main Battery CAGE CODE 63017	DWG NO. 5-0142	REV H
	SHEET 3 of 10	

- 1. Scope: This Maintenance Manual Supplement provides the additional data required to insure satisfactory operation, maintenance, and repair of the Concorde valve regulated lead acid battery installation.
- 2. Purpose: This manual sets forth the instructions for determining continued airworthiness of a Concorde valve regulated lead acid battery.
- 3. Application: Concorde valve regulated aircraft batteries RG series.

4. Definitions:

- a. Valve regulated battery A lead acid battery in which there is no free electrolyte. This battery requires no maintenance of the liquid level and recombines the gases formed on charge within the battery. The battery may be used in any attitude without danger of leakage or spilling of electrolyte.
- b. Rated capacity C1 Quantity of electricity in Ampere-hours (Ah) which the cell or battery is capable of delivering in 1 hr.
- c. End Point Voltage (EPV) Unless otherwise stated, during discharge the battery voltage corresponding to a mean voltage per cell of 1.67 Volts for lead-acid batteries. (10 EPV for 12 volt and 20 EPV for 24 volt batteries).

5. Precautions:

- a. CAUTION: Aircraft batteries are certified to have certain minimum capacity for emergency operations in the event of a electrical generator system failure. Never "Jump Start" an aircraft that has a discharged or "Dead" battery.
- b. WARNING: ELECTRIC SHOCK HAZARD. Do not touch uninsulated portion of the connector or the battery terminals. A possibility of serious electrical shock exists.
- c. WARNING: ELECTRIC SHOCK HAZARD. Do not lay tools or other metal objects on the battery as arcing or explosion could occur. Remove conductive jewelry before working around battery, charger, or test equipment.
- d. CAUTION: ELECTRIC BURN HAZARD. Do not wear conductive rings, belt buckles, or other jewelry when working with batteries, chargers, or test equipment. Do not lay tools or other metal objects on the battery as arcing and severe burns could occur.
- e. **WARNING:** Batteries on charge or discharge produce hydrogen gas, which can explode if ignited. Do not smoke, use an open flame, or cause sparking near a battery. Charge, service or test a battery only in a well ventilated area. The use of exhaust fans may reduce the risk of explosion.
- f. WARNING: Batteries contain sulfuric acid which will cause burns. DO NOT TOUCH EYES AFTER TOUCHING BATTERY. Do not get acid in your eyes, or on your skin, or clothing. In the event of acid in the eyes, flush thoroughly with clean cool water for several minutes. Get professional medical attention. Refer to battery MSDS for additional information.

Instructions for Continued Airworthiness, Concorde Valve Regulated Lead Acid Main Battery	5-0142 REV
Concorde Valve Regulated Load / total	SHEET 4 of 10
CAGE CODE 63017	

- WARNING: Wear proper eye, face and hand protection at all times when
- WARNING: Wear proper eye, lace distribution and use of emergency eyewash and working with batteries. Know the location and use of emergency eyewash and shower nearest the battery charging area.
- h. CAUTION: To prevent damage to the connector, arc burns, or explosion. batteries should never be connected or disconnected while being charged or discharged. Batteries must be connected or disconnected only when the circuit is open. Ensure the aircraft battery switch, external power source, or the charger analyzer is in the OFF position before connecting or disconnecting the battery Battery terminal protectors should be installed whenever the battery is not connected in the aircraft or to the test equipment.
- CAUTION: Batteries contain hazardous materials. Know the location and proper use of emergency response materials. Refer to battery Material Safety Data Sheet (MSDS) for additional information.
- Caution / Warning: Only constant potential charging may be done on the aircraft. DO NOT constant current charge a battery on the aircraft. There may be a serious risk of injury to personnel and / or damage to the aircraft or aircraft systems due to high voltage and generation of explosive gases when charging constant current.
- 6. Airworthiness Limitations: There are no airworthiness limitations associated with the installation of a Concorde valve regulated lead acid battery in an aircraft.
- 7. Installation and Removal:

NOTE: The following instructions are generic. See airframe manufacturer's maintenance manuals or STC for instructions specific to a particular aircraft model. For removal and replacement of a nickel cadmium battery, see airframe maintenance manual or STC.

- a. Installation procedure:
 - Removal:
 - Set Master Switch to the OFF position.
 - Disconnect any external power supply.
 - Open battery compartment access panels.
 - Disconnect battery quick disconnect plug or remove terminal bolts and disconnect battery cables from battery terminals. Always disconnect the ground cable first and install the ground cable last.

 - Disconnect battery ventilation tubes, if any.
 - Unlock battery hold down clamps or remove battery hold down
 - Carefully remove battery.

WARNING: Batteries are heavy. Use appropriate lifting devices or equipment. Use battery handles where provided.

Inspect the battery for damage. Cracks in metal or plastic containers are not permitted. Dents in metal containers that

Concorde Valve Regulated Lead Acid Markets		070
Cage Code 63017 Concorde Valve Regulated Lead Acid Main Battery	DWG NO. 5-0142	REV H
	SHEET 5 of 10	

- impinge on the interior plastic container are not acceptable.
- Set Master Switch to the OFF position. (2)
- Disconnect any external power supply.
- Open battery compartment access panels.
- Ensure the battery container or tray is clean and dry.
- Install battery in battery container or tray.

WARNING: Batteries are heavy. Use appropriate lifting devices or equipment. Use battery handles where provided.

- Engage battery hold down hardware, torque and safety wire per airframe manufacturer maintenance manual.
- Connect battery vent tubes to aircraft ventilation system, if any.
- Connect battery quick disconnect plug, any auxiliary connector or for ring terminals, install with bolt and bevel lock washer provided with the battery. Torque terminal bolts as noted on the battery label. Always disconnect the ground cable first and install the ground cable last.
- CAUTION: Use an open end wrench on the flats of the battery terminal, where available, while torquing the terminal bolts. Failure to do so may result in the rupture of the battery seal at the terminal and premature failure of the battery.
- CAUTION: Use only the hardware provided with the battery. Do not use stainless steel or steel washers between the ring terminal and the battery terminal.
 - Replace electrical compartment access panel.
 - Update aircraft weight and balance data, if necessary.
 - Perform an operational test. (12)
 - (13) Annotate log book with battery serial number and date of installation.
- 8. Inspection Requirements and Overhaul Schedule:
 - a. Scheduled inspections:

NOTE: If the battery is used to satisfy the essential power requirement, it must be capacity tested. If the battery is not used for essential power, there is no requirement for periodic capacity checks.

- If the battery is being used to start a turbine APU or a reciprocating engine, the battery can be operated for 24 months or 1200 hours of operation whichever occurs first before the initial capacity check. Subsequent capacity checks are due each additional 12 months or 200 hours of operation, whichever occurs first.
- If the battery is used to start a Turbine engine, regardless of whether or not it is also being used to start a turbine APU, the initial capacity check is at 12 months or 600 hours of operation, whichever occurs first. Subsequent capacity checks are due each additional 3 months or 200 hours of operation whichever occurs first.

Instructions for Continued Airworthiness, Concorde Valve Regulated Lead Acid Main Battery	DWG NO. 5-0142	REV H
Concorde Valve Regulated Lead / Isla Main	SHEET 6 of 10	
CAGE CODE 63017		

Two (2) or more batteries operated in parallel should be replaced together for maximum reliability. Replacing only one battery in a parallel string for maximum reliability. Replacing only the resistance of the new battery results in unequal charge acceptance as the resistance of the older battery. is much different than the resistance of the older battery.

NOTE: The initial recommended inspection schedule may be adjusted to NOTE: The initial recommended inspectation inspection interval so long as coincide with the airframe or engine manufacturer's inspection interval so long as the operating time does not exceed the above limits by more than 10% or 60 hours. The inspection schedule may be adjusted after the useful battery life is established in a particular operation. After the useful life of a battery is established for a particular aircraft or operating mode, the actual useful life period in months or hours may be substituted for the above schedule.

- b. Non-scheduled Inspections:
 - Capacity check if abnormal slow engine starting performance is noted.
 - Capacity check if abnormal high charging current is required to maintain the battery at buss voltage.
- c. Inspection Procedure:
 - Charge the battery:
 - (1) Special tools:
 - Advanced Power Products Beta D-50 Aircraft Battery Analyzer, P/N 4126, or equal.
 - Advanced Power Products Alpha C-25 Battery Charger, P/N 4142, or equal.
 - Advanced Power Products CA-1550 Charger / Analyzer P/N 4159, or equal.

Warning: Contact Concorde for determination of equal test equipment. Some brands of battery chargers will destroy the

Depending on the type of charger available, charge the battery Constant Potential (CP). Charge at 14.1 volts for 12 volt batteries or 28.2 volts for 24 volt batteries until the charge current stabilizes for 1 hour.

Caution / Warning: Only constant potential charging may be done on the aircraft. DO NOT constant current charge a battery on the aircraft. There may be a serious risk of injury to personnel and / or damage to the aircraft or aircraft systems.

Capacity test:

Stabilize the battery at 15 °C (59 °F) or higher. The battery must be at the temperature for at least 24 hours.

Discharge the battery at the C1 rate on the label to an end point voltage of 10 volts for 12 volt batteries or 20 volts for 24 volt batteries, or the rate and end point voltage (EPV) specified by the airframe manufacturer for essential power.

Record the time to EPV.

The battery is acceptable for continued use if the ampere hour

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capacity (actual hours of discharge x ampere rate of discharge) is greater than 85% of the nominal rated capacity (C1) shown on the label (i.e. 51 minutes or more). If the battery passes the capacity test return it to service. If the battery fails the capacity test perform a conditioning procedure (13. c.). After the battery has been conditioned perform a second capacity test. If the battery passes return it to service. If the battery fails replace it.

NOTE: Airframe or accessory equipment manufacturers may specify a different capacity requirement.

- Return to service: Charge the battery as above. If the battery gets very hot (external case temperature greater than 55 • C / 130 • F) during charging, it should be replaced.
- d. Component Overhaul Schedule: No component overhaul required for this type product.

9. Troubleshooting:

Symptom	Probable Cause	Corrective Action
Low voltage / no voltage	Battery partially or fully discharged	Charge in accordance with Section 8
Battery does not hold charge	Battery beyond serviceable life	Replace battery
Battery gets hot while recharging	Battery beyond serviceable life	Replace battery

Servicing discharged batteries:

a. Recharging dual (parallel) batteries on the aircraft should be done individually.

Recharging a single battery on the aircraft:

Special tools:

Advanced Power Products Activator 282 Battery Charger, P/N 4105, or equal constant potential charger.

The Activator 282 is fully automatic. Connect to battery and initiate charging. If using a constant potential charger, follow instructions below.

Caution / Warning: Only constant potential charging may be done on the aircraft. Ensure the charging area is well ventilated. DO NOT constant current charge a battery on the aircraft. There may be a serious risk of injury to personnel and / or damage to the aircraft and aircraft systems.

Warning: Contact Concorde for determination of equal test equipment. Some brands of battery chargers will destroy the battery.

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Set the charger for 2.35 volts per cell (14.1 volts for a 12 volt battery, 28.2

volts for a 24 volt battery).

Set the current limit to the C1 rate. NOTE: If current on external power unit is not adjustable, do not charge the

battery in the aircraft.

Initiate charging and continue to charge until the charge current stabilizes for 1 hour.

c. Uninstalled recharging:

Remove the battery from the aircraft.

Special tools:

(1) Advanced Power Products Alpha C-25 Battery Charger, P/N 4142: Advanced Power Products CA-1550 Charger / Analyzer P/N 4159:

Warning: Contact Concorde for determination of equal test equipment. Some brands of battery chargers will destroy the

Depending on the type of charger available, charge the battery Constant Potential (CP). Charge at 14.1 volts for 12 volt batteries or 28.2 volts for 24 volt batteries until the charge current stabilizes for 1 hour.

NOTE: Batteries that have been allowed to stand in a deeply discharged state may not accept a CP recharge. See conditioning procedure in paragraph 13.c for handling those batteries.

11. Repair / Replacement:

- a. Repairs should be performed only by a Concorde approved battery shop.
- b. Replacement may be made by removing and installing a new battery in accordance with the instructions in this supplement.
- c. The battery should be replaced after 3 years or 1800 hours of operation or whichever occurs first.
- 12. Facilities: Valve regulated batteries may be serviced in any battery facility, including nickel cadmium service facilities. The battery is sealed to prevent cross contamination of the electrolyte.
- 13. Storage Limitations:
 - a. Batteries are serviced and charged at the factory prior to shipment.

i.For maximum battery life, boost charge when open circuit voltage is below 25.0

volts for 24 volt batteries and 12.5 volts for 12 volt batteries

ii.Batteries that have not been recharged when stored for long periods are to be conditioned (13.c) and tested (8.c) before being placed in service. b. Boost charging procedure:

i.Special tools:

Advanced Power Products Alpha C-25 Battery Charger, P/N 4142; Advanced Power Products Activator 282 Battery Charger, P/N

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4105; Advanced Power Products CA-1550 Charger / Analyzer, P/N 4159; or equal.

ii. Depending on the type of charger available, charge the battery Constant Potential (CP). Charge at 14.1 volts for 12 volt batteries or 28.2 volts for 24 volt batteries until the charge current stabilizes for 1 hour.

c. Conditioning procedure:

Warning: The battery must be removed from the aircraft prior to performing a conditioning charge.

i.Special tools:

- Advanced Power Products Beta D-50 Aircraft Battery Analyzer P/N 1. 4146 or 4159, or equal.
- Advanced Power Products Alpha C-25 Battery Charger P/N 4142 or 4159, or equal. Warning: Contact Concorde for determination of equal test equipment. Some brands of battery chargers will destroy the

ii.Procedure:

- Discharge the battery at the C1 rate to an end point voltage of 9 1. volts for 12 volt batteries and 18 volts for 24 volt batteries.
- Constant current charge at the 1 /10 of the C1 rate for 16 hours. WARNING: This procedure may damage the battery if performed on a repetitive basis.

Allow the battery to cool down for 8 hours.

Retest the battery following the inspection procedure (8.c). 4.

14. Disposal:

- a. Batteries contain lead, sulfuric acid, and other hazardous materials. Dispose of all spent batteries in accordance with local laws and regulations. See battery Material Safety Data Sheet (MSDS) for additional information.
- b. Lead acid batteries are recyclable.

battery.

CAUTION: Some aircraft batteries are encased in aluminum containers. These containers must be removed prior to recycling.

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INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

MAINTENANCE MANUAL SUPPLEMENT

CONCORDE FLOODED LEAD ACID MAIN BATTERY

This document must be used in conjunction with the basic Maintenance Manual for the aircraft when the Concorde Lead Acid Battery is installed. This Maintenance Manual Supplement modifies/augments the portions of the basic Maintenance Manual.

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RECORD OF REVISIONS

Rev. No.	Description	Date	Appd
А	Corrected page number Corrected metric torque Deleted paragraph a, recharging on the aircraft	11/28/00	JBT
В	Add note on watering Add hydrometer & SG readings	11/30/01	JBT
С	Revised format Assigned drawing number Minor text revisions Add notes on inspection requirement	6/20/02	JBT
D	Completely revised	7/3/03	JBT
Е	Rev activation instructions Delete taper and CP charging Rev CP charging voltage	10/19/04	JBT

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- Scope: This Maintenance Manual Supplement provides the additional data required to insure satisfactory operation, maintenance, and repair of the Concorde flooded lead acid battery installationScope: This Maintenance Manual Supplement provides the additional data required to insure satisfactory operation, maintenance, and repair of the Concorde flooded lead-acid battery installation.Scope: This Maintenance Manual Supplement provides the additional data required to insure satisfactory operation, maintenance, and repair of the Concorde flooded lead-acid battery installation.
- Purpose: This manual sets forth the instructions for determining continued airworthiness of a Concorde flooded lead acid battery.
- Application: Concorde dry charged (flooded) aircraft batteries CB series. 3.

4. Definitions:

- Flooded battery A lead acid battery that contains liquid electrolyte. For aircraft, these are equipped with aerobatic vent caps to prevent the spilling of acid during maneuvers requiring a change in attitude of the battery.
- Rated capacity C1 Quantity of electricity in Ampere-hours (Ah) which the cell or battery is capable of delivering in 1 h, throughout its normal service life, after full charge, under conditions defined with regard to temperature and end discharge voltage.
- End Point Voltage (EPV) Unless otherwise stated, during discharge the battery voltage corresponding to a mean voltage per cell of 1.67 Volts for lead-acid batteries.

5. Precautions:

- CAUTION: Aircraft batteries are certified to have certain minimum capacity for emergency operations in the event of a electrical generator system failure. Never "Jump Start" an aircraft that has a discharged or "Dead" battery.
- WARNING: ÉLECTRIC SHOCK HAZARD. Do not touch uninsulated portion of the connector or the battery terminals. A possibility of serious electrical shock exists.
- WARNING: ELECTRIC SHOCK HAZARD. Do not lay tools or other metal objects on the battery as arcing or explosion could occur. Remove conductive jewelry before working around battery, charger, or test
- CAUTION: ELECTRIC BURN HAZARD. Do not wear conductive rings, belt buckles, or other jewelry when working with batteries, chargers, or test equipment. Do not lay tools or other metal objects on the battery as arcing and severe burns could occur. WARNING: Batteries on charge or discharge produce hydrogen gas,

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- which can explode if ignited. Do not smoke, use an open flame, or cause sparking near a battery. Charge, service or test a battery only in a well ventilated area. The use of exhaust fans may reduce the risk of explosion. WARNING: Batteries contain sulfuric acid which will cause burns. DO
- NOT TOUCH EYES AFTER TOUCHING BATTERY. Do not get acid in your eyes, or on your skin, or clothing. In the event of acid in the eyes, flush thoroughly with clean cool water for several minutes. Get professional medical attention. Refer to battery MSDS for additional information.
- WARNING: Wear proper eye, face and hand protection at all times when g. working with batteries. Know the location and use of emergency eyewash and shower nearest the battery charging area.
- CAUTION: To prevent damage to the connector, arc burns, or explosion. batteries should never be connected or disconnected while being charged or discharged. Batteries must be connected or disconnected only when the circuit is open. Ensure the aircraft battery switch, external power source, or the charger/analyzer is in the OFF position before connecting or disconnecting the battery. Battery terminal protectors should be installed whenever the battery is not connected in the aircraft or to the test equipment.
- CAUTION: Batteries contain hazardous materials. Know the location and proper use of emergency response materials. Refer to battery Material Safety Data Sheet (MSDS) for additional information.
- Caution / Warning: Only constant potential charging may be done on the aircraft. DO NOT constant current charge a battery on the aircraft. There may be a serious risk of injury to personnel and / or damage to the aircraft or aircraft systems due to high voltage and generation of explosive gases when charging constant current.
- Airworthiness Limitations: There are no airworthiness limitations associated with the installation of a Concorde flooded lead acid battery in an aircraft.
- Activation of Dry Charged Batteries:
 - Remove the sealing tape from the vents and remove the vent caps.
 - Fill each cell with 1.285 SG electrolyte to a height just above the top of the
 - Let the battery stand until it cools to room temperature or for one hour. Then, add more electrolyte to 1/4 inch above the top of the plates and
 - Install vent caps.
 - Boost charge the battery Constant Current (CI) charge at 1/10 of the C1

NOTE: The battery is fully charged when the voltage stabilizes or decreases slightly and the SG of the electrolyte stabilizes for three

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successive readings taken at one hour intervals.

WARNING: Complete charging after activation is required for satisfactory performance. Do not allow the battery temperature to exceed 120° F.

After the battery is fully charged, it may be necessary to adjust the electrolyte SG of each cell to 1.285 by diluting higher SG cells with approved water and by filling the lower SG cells with stronger electrolyte.

NOTE: Add water only when the battery is warm, such as immediately after charge or flight. Do not overfill.

- Wash and dry outside of battery before installation.
- Installation and Removal:

NOTE: The following instructions are generic. See airframe manufacturer's maintenance manuals or STC for instructions specific to a particular aircraft model. For removal and replacement of a nickel cadmium battery, see airframe maintenance manual or STC.

- Installation procedure:
 - Remove old battery:
 - Set Master Switch to the OFF position.
 - Disconnect any external power supply
 - Open battery compartment access panels.
 - Disconnect battery quick disconnect plug or remove terminal bolts and disconnect battery cables from battery terminals. Always disconnect the ground cable first.
 - Disconnect battery ventilation tubes, if any.
 - Unlock battery hold down clamps or remove battery hold down bars. Disengage battery.
 - Carefully remove battery.

WARNING: Batteries are heavy. Use appropriate lifting devices or equipment. Use battery handles where provided. Install new battery:

- Set Master Switch to the OFF position.
- Disconnect any external power supply.
- Open battery compartment access panels. Ensure the battery container or tray is clean and dry.
- Install battery in battery container or tray.
- WARNING: Batteries are heavy. Use appropriate lifting devices or

Use battery handles where provided. Engage battery hold down hardware, torque and safety wire equipment.

- per airframe manufacturer maintenance manual. Connect battery vent tubes to aircraft ventilation system, if
- Connect battery quick disconnect plug, or for ring terminals,
- install with bolt and bevel lock washer provided with the

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battery. Torque terminal bolts as noted on the battery label. Always install the ground cable last.

CAUTION: Use an open end wrench on the flats of the battery terminal while torquing the terminal bolts. Failure to do so may result in the rupture of the battery seal at the terminal and premature failure of the battery. CAUTION: Use only the hardware provided with the battery. Do not place any stainless steel or steel washers between the ring terminal and the battery terminal.

- Replace electrical compartment access panel. (9)
- (10) Update aircraft weight and balance data, if necessary.
- (11) Perform an operational test.
- (12) Annotate log book with battery serial number and date of installation.
- Inspection Requirements and Overhaul Schedule:
 - Inspection Requirements:

NOTE: If the battery is used to satisfy the essential power requirement, it must be capacity tested. If the battery is not used for essential power, there is no requirement for periodic capacity checks.

- Scheduled inspections:
 - An initial check of the emergency capacity is required 12 months after installation or after 600 hours of operation, whichever occurs first.
 - After the initial check, a check of emergency capacity is required after 12 months or after every additional 200 hours of operation.

NOTE: The inspection schedule may be adjusted after the useful battery life is established in a particular operation. After the useful life of a battery is established for a particular aircraft or operating mode, the actual useful life period in months or hours may be substituted for the above schedule.

- Inspection Procedure:
 - Charge the battery:
 - Special tools:
 - Advanced Power Products Beta D-50 Aircraft Battery Analyzer, P/N 4126 or equal.
 - Advanced Power Products Alpha C-25 Battery Charger, P/N 4142 or Advanced Power Products Activator 282 Battery Charger, P/N 4105 or equal.
 - Advanced Power Products CA-1550 Charger / Analyzer P/N 4159 or equal.
 - Hydrometer P/N 9295 or equal. Depending on the type of charger available, charge the battery Constant Potential (CP). Charge at 14.4 volts for 12

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volt batteries or 28.8 volts for 24 volt batteries until the charge current stabilizes for 1 hour.

The battery is fully charged when the voltage stabilizes or decreases slightly and the SG of the electrolyte stabilizes for three consecutive successive readings taken at one hour

Capacity test:

Stabilize the battery at 15°C (59°F) or higher. The battery must be at the temperature for at least 24 hours.

Discharge the battery at the C1 rate to an end point voltage of 10 volts for 12 volt batteries or 20 volts for 24 volt batteries, or the rate and end point voltage (EPV) specified by the airframe manufacturer for essential power.

Record the time to EPV.

The battery is acceptable for continued use if the ampere hour capacity (actual hours of discharge x ampere rate of discharge) is greater than 85% of the nominal rated capacity (C1) shown on the label (i.e. 51 minutes or more). If the battery passes return it to service. If the battery fails replace

NOTE: Airframe or accessory equipment manufacturers may specify a higher capacity requirement.

If the battery gets very hot (external case temperature greater than 55°C / 130°F) during charging, it should be replaced.

Overhaul Schedule: No component overhaul required for this type product.

10. Troubleshooting:

Symptom	Probable Cause	Corrective Action
Low voltage / no voltage	Battery partially or fully discharged	Charge in accordance with Section
Battery does not hold charge	Battery beyond serviceable life	Replace battery Replace battery
Battery gets hot while recharging	Battery beyond serviceable life	Replace buttony

Recharging dual (parallel) batteries on the aircraft should be done 11. Servicing discharged batteries:

Recharging a single battery on the aircraft:

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Advanced Power Products Activator 282 Battery Charger, Special tools:

P/N 4105, or equal constant potential charger. The Activator 282 is fully automatic. Connect to battery and

initiate charging. If using a constant potential charger, follow instructions below.

Caution / Warning: Only constant potential charging may be done on the aircraft. Ensure the charging area is well ventilated. DO NOT constant current charge a battery on the aircraft. There may be a serious risk of injury to personnel and / or damage to the aircraft and aircraft systems.

Set the charger for 2.4 volts per cell (14.4 volts for a 12 volt battery. 28.8 volts for a 24 volt battery).

Set the current limit to the C1 rate.

NOTE: If current on external power unit is not adjustable, do not charge the battery in the aircraft

- The battery is fully charged when the voltage stabilizes or decreases slightly and the SG of the electrolyte stabilizes for three consecutive successive readings taken at one hour intervals.
- Uninstalled recharging: C.
 - Remove the battery from the aircraft.
 - Special tools:
 - Advanced Power Products Alpha C-25 Battery Charger, P/N 4142; Advanced Power Products CA-1550 Charger / Analyzer P/N 4159; or equal.
 - Depending on the type of charger available, charge the battery Constant Potential (CP), Constant Current (CI), or with a taper charger
 - If Constant Potential, charge at 14.4 volts for 12 volt batteries or 28.8 volts for 24 volt batteries until the charge current stabilizes for 1 hour.

NOTE: Batteries that have been allowed to stand in a deeply discharged state may not accept a CP recharge. See conditioning procedure in paragraph 14.c for handling those batteries.

If Constant Current, the maximum charge rate in Amperes is the C1 rate on the label divided by 3.

Warning: Constant current charging must be monitored. Charge current must be reduced or charging discontinued when the battery gets hot. Electrolyte lost due to charging must be replaced with water.

The battery is fully charged when the voltage stabilizes or decreases slightly and the SG of the electrolyte stabilizes for three consecutive successive readings taken at one hour intervals.

Replacement / Repair:

Repairs should be performed only by a Concorde approved battery shop.

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Replacement may be made by removing and installing a new battery in b. accordance with the instructions in this supplement.

Facilities: 13.

Flooded lead-acid batteries must be serviced in a shop designated for a.

Storage Limitations: 14.

- Batteries should be stored in a cool and dry place.
- Dry charged batteries may be stored for an indefinite period prior to b.
- After activation,
 - For maximum battery life, boost charge when open circuit voltage is below 25.0 volts for 24 volt batteries and 12.5 volts for 12 volt batteries.
 - Batteries that have not been recharged when stored for long periods are to be conditioned (14.e) and tested (9.b.ii) before being placed in service.
- Boost charging procedure: d.
 - Special tools:
 - Advanced Power Products Alpha C-25 Battery Charger, P/N 4142; Advanced Power Products Activator 282 Battery Charger, P/N 4105; Advanced Power Products CA-1550 Charger / Analyzer, P/N 4159; or equal.
 - Depending on the type of charger available, charge the battery Constant Potential (CP). Charge at 14.4 volts for 12 volt batteries or 28.8 volts for 24 volt batteries until the charge current stabilizes for 1 hour.
- Conditioning procedure: e.

Warning: The battery must be removed from the aircraft prior to performing a conditioning charge.

- Special tools:
- Advanced Power Products Beta D-50 Aircraft Battery Analyzer P/N 4126 or equal.
 - Advanced Power Products Alpha C-25 Battery Charger P/N 4142 or equal.
- Procedure: ii.

- Discharge the battery at the C1 rate to an end point voltage of 9 volts for 12 volt batteries and 18 volts for 24 volt Constant current charge at 1 /10 of the C1 rate for 16 hours.

WARNING: This procedure may damage the battery if performed on djust the electrolyte level by adding approved water only. a repetitive basis.

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- Allow the battery to cool down for 8 hours.

 Test the battery following the inspection procedure (9.b.ii).
- 15. Disposal:
- Batteries contain lead, sulfuric acid, and other hazardous materials.
 - Lead acid batteries are recyclable. CAUTION: Some aircraft batteries are encased in aluminum containers. These containers must be removed prior to recycling.
 - Dispose of all spent batteries in accordance with local laws and regulations. See battery Material Safety Data Sheet (MSDS) for additional information.

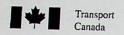
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Aircraft Certification Division 800-1601 Airport Road NE Calgary, Alberta, Canada T2E 678

August 26, 2008

Your file Votre reference

File: NAPA C-08-0724

Phone: 403-292-4145 FAX: 403-292-4992 email: hoeppnd@tc.gc.ca

Concorde Battery Corp. 2009 San Bernardino Rd. West Covina, CA USA 919790

Attention: Ramuntxo Duhart

Subject: Notification of Transport Canada Acceptance of FAA STC SA01147WI Aircraft: Piper PA-25 Pawnee Series and PA-36 Pawnee Brave Series Agricultural Aircraft per Canadian Type Certificates A-118 and A-228

Concorde FLA or VRSLA Series Batteries in accordance with Concorde Battery Corporation Master Drawing List No. 5-0157, Rev A or later FAA Approved Revision

This will notify you that Transport Canada has reviewed and accepted FAA STC SA01147WI in respect to the PA-25 and PA-36 series agricultural aircraft in accordance with the Bilateral Aviation Safety Agreement between the US and Canada without the issue of a corresponding Canadian STC.

All other aircraft types listed in the Approved Model List (AML) are automatically accepted in Canada without review because they are small (CAR 3 / FAR 23), nonrestricted category US State of Design aircraft covered by a US STC.

This STC will be added to the Transport Canada listing of FAA STC's which have been accepted or familiarized in Canada. This listing is available on the TC Web Site at: http://www.tc.gc.ca/aviation/applications/nico-celn/en/adv_search.asp?x_lang=e

This letter confirms formal acceptance of the referenced STC by Transport Canada.

Dennis Hoeppner, P. Eng.

Senior Engineer - Aircraft Certification

CC:

Federal Aviation Administration Wichita Aircraft Certification Office 1801 Airport Road, Room 100 Wichita, Kansas USA 67209

Attention - Harvey Nero, Program Manager

Canadä